

Weight Shift & %MAC Worksheet

This worksheet has two parts: at the left is the weight shift function and at the %MAC function.

Weight Shift & %MAC
Clear

<div style="border: 1px solid black; background-color: #0056b3; color: white; padding: 5px; margin-bottom: 5px;"> Total Weight Wt = 7,500 LB </div> <div style="border: 1px solid black; background-color: #0056b3; color: white; padding: 5px; margin-bottom: 5px;"> C.G. Change ΔCG = 1.00 IN </div> <div style="border: 1px solid black; background-color: #0056b3; color: white; padding: 5px; margin-bottom: 5px;"> Arm Change ΔArm = 120.00 IN </div> <div style="border: 1px solid black; background-color: #800000; color: white; padding: 5px;"> Weight to Shift ΔWt = 62.5 LB </div>	<div style="border: 1px solid black; background-color: #0056b3; color: white; padding: 5px; margin-bottom: 5px;"> Center of Gravity CG = 910.20 IN </div> <div style="border: 1px solid black; background-color: #0056b3; color: white; padding: 5px; margin-bottom: 5px;"> Mean Aerod. Chor MAC = 180.70 IN </div> <div style="border: 1px solid black; background-color: #0056b3; color: white; padding: 5px; margin-bottom: 5px;"> Leading Edge of MAC LMAC = 860.20 IN </div> <div style="border: 1px solid black; background-color: #800000; color: white; padding: 5px;"> Percent of MAC %MAC = 27.7 % </div>
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Weight Shift Calculations

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%MAC Calculations

Clear	Set all variables to a invalid state keeping the current value. If it is touched again, clears all values to 0.		
Weight Shift		%MAC	
Wt	Total weight of the aircraft.	CG	Center of gravity Arm of the aircraft..
ΔCG	Change in the center of gravity arm	MAC	Mean aerodynamic chop length.
ΔArm	Change in the arm of the weight to shift.	LMAC	Leading edge arm of the MAC.
ΔWt	Weight to shift to new location.	%MAC	Percentage of MAC.

The **Weight Shift** function computes the amount of weight that must shift to move the CG by a desired amount. Also, can be used to find the change in CG or Arm from adding or removing weight.

The **%MAC** function calculates the %MAC given the CG, the length of the mean aerodynamic chord (MAC), and the leading edge of the mean aerodynamic chord (LMAC).

NOTE: Always verify the physical units

To change the units of a variable, tap over the unit symbol and select the right one from the pop-up menu. To change the whole units in the worksheet select “Set Metric Units” or “Set US Units” from the [**UNITS▶**] button.

All the following examples use US units. So please select “Set US Units” from the [**UNITS▶**] menu in the Navigation Bar.

Example 1:

Find weight of the item that must shift to move the CG in 1 IN if the total weight is 7,500 LBS and the distance weight is shifted is 120 IN.

Solution:

Keystrokes	Description
[Clear]	Clears all variables to start a new calculation and Leg-1 selected.
type 7500 [Wt]	Stores 7,500 LB in the aircraft weight, Wt
type 1 [ΔCG]	Stores 1 IN of change in the center of gravity, ΔCG
type 120 [ΔArm]	Stores 120 IN change in the arm to shift the weight, ΔArm and automatically show the calculations weight to shift: ΔWt = 62.5 LB

Example 2:

Determine the CG in percent of MAC if MAC extends from 860.2 to 1040.9 inches and the center of gravity is at 910.2 inches.

Solution:

Keystrokes	Description
type 860.2 [LMAC]	Stores 860.2 IN in the left edge of MAC, LMAC
[+/-] [+] type 1040.9 [MAC]	Stores 180.7 IN in mean aerodynamic chor, MAC
type 910.2 [CG]	Stores 910.2 IN in center of gravity arm, CG and automatically calculates: %MAC = 27.7 %

Appendix : Equations Used

The equations that this worksheet calculates are:

Weight Shift:

a) Center of Gravity Change, **ΔCG** :

$$\mathbf{\Delta CG} = \Delta Wt \cdot \Delta Arm / Wt$$

b) Item Weight Arm Change, **ΔArm** :

$$\mathbf{\Delta Arm} = Wt \cdot \Delta CG / \Delta Wt$$

c) Item Weight to shift, **ΔWt** :

$$\mathbf{\Delta Wt} = Wt \cdot \Delta CG / \Delta Arm$$

%Mach:

a) Mean Aerodynamic Chor:

$$\mathbf{MAC} = (CG - LMAC) \cdot 100 / \%MAC$$

b) Center of Gravity Arm:

$$\mathbf{CG} = MAC \cdot \%MAC / 100 + LMAC$$

c) Left Edge of MAC Arm:

$$\mathbf{LMAC} = CG - MAC \cdot \%MAC / 100$$

d) Center of Gravity Percent of MAC:

$$\mathbf{\%MAC} = 100 \cdot (CG - LMAC) / MAC$$